

**VOLUME I  
PERFORMANCE FLIGHT TESTING**

**APPENDIX A  
SYMBOLS  
TERMS AND ABBREVIATIONS**

*1982*

**DTIC QUALITY INSPECTED 4**

**19970116 075**

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## SYMBOLS, TERMS, AND ABBREVIATIONS

ARABIC <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
a	Acceleration	ft/sec <sup>2</sup>
a	Lift curve slope	per deg or per rad
a	Speed of sound	ft/sec, mi/hr, kts
ac	Aerodynamic center	
A	Area	ft <sup>2</sup> , m <sup>2</sup>
AR	Aspect ratio	
b	Wingspan	ft, m
	Blade Width	ft, m
B	Number of blades	
BHP	Brake horsepower	
B.L.	Base line	
c	Absolute velocity	
c	Chord	ft, m
c	Compression	
c	Specific fuel consumption	lb/hr
°C	Degrees centigrade	deg

SYMBOLS, TERMS, AND ABBREVIATIONS

<u>ARABIC Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$c_r$	Root chord	ft, m
$c_t$	Tip chord	ft, m
$c_p$	Specific heat at constant pressure	btu/lb $^{\circ}$ R
$c_v$	Specific heat at constant volume	btu/lb $^{\circ}$ R
$c_d$	Section drag coefficient	
$c_f$	Skin friction coefficient	
$c_l$	Section lift coefficient	
$c_m$	Section moment coefficient	
$c_F$	Force coefficient	
$c_D$	Aircraft drag coefficient	
$c_L$	Aircraft lift coefficient	
$c_{L_{ic}}$	Indicated lift coefficient	
$c_M$	Aircraft moment coefficient	
$c_p$	Pressure coefficient	

SYMBOLS, TERMS, AND ABBREVIATIONS

ARABIC <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$C_P$	Propeller power coefficient	
$C_Q$	Propeller torque coefficient	
$C_T$	Propeller thrust coefficient	
$cg$	Center of gravity	
$cp$	Center of pressure	
CR	Compression ratio	
CPR	Compressor Pressure ratio	
d	Differential	
D	Diameter	ft
D	Drag	lb
D	Diffuser	
$d/dt$	Time rate of change	
$dC_L/d\alpha$	Lift curve slope	per deg or per rad
e	Oswald's efficiency factor	
E	Shear modulus	

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>ARABIC Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
E	Endurance	hr
E	Total energy	ft lbs
$E_m$	Maneuver energy	ft lbs
$E_s$	Specific energy	ft
EGT	Exhaust gas temperature	deg
f	Function of	
f	Equivalent flat plate area	ft <sup>2</sup>
F	Force	lb
F	Fan	
F	Resultant aerodynamic force	lb
$^{\circ}F$	Degrees Fahrenheit	deg
$F_g$	Gross thrust	lb
$F_n$	Net thrust	lb
$F_{ex}$	Excess thrust	lb
F.R.L	Fuselage reference line	

SYMBOLS, TERMS, AND ABBREVIATIONS

<u>ARABIC</u> <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
F.S.	Fuselage station	
g	Acceleration due to gravity	ft/sec <sup>2</sup>
G	Gravitational constant	32.17405 ft <sup>2</sup> /sec <sup>2</sup> geo- potential ft
h	Enthalpy	btu/lb
h	Tapeline altitude	ft
h <sub>v</sub>	Kinetic energy	
H	Total head pressure	lb/in <sup>2</sup>
H	Combustor	
H	Altitude, general	ft
H	Geopotential at a point	geopotential ft
H <sub>c</sub>	Pressure altitude	ft
H <sub>i</sub>	Indicated altitude	ft

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>ARABIC</u> <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$H_{ic}$	Indicated altitude corrected for instrument error, $H_{ic} + \Delta H_{ic}$	ft
$\Delta H_{ic}$	Altimeter instrument correction	ft
$H_{ic_l}$	Indicated altitude corrected for instrument and lag errors, $H_i + \Delta H_{ic} + \Delta H_{ic_l}$	ft
$\Delta H_{ic_l}$	Altimeter lag correction	ft
$\Delta H_p$	Altimeter position error corresponding to $\Delta P_p$	ft
$\Delta H_{pc}$	Altimeter position error correction	ft
HP	Horsepower	hp
H.V.	Heating value of hydrocarbon fuel	btu/lb
$I_s$	Specific impulse	sec
J	Propeller advance ratio	
$K_n$	A constant	
$K_t$	Temperature probe recovery factor	
$^oK$	Degrees Kelvin	deg

## SYMBOLS, TERMS, AND ABBREVIATIONS

ARABIC <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
KE	Kinetic Energy	
$l$	Characteristic length	ft
ln	Natural logarithm	
L	Lift	lb
L	Length, dimensional analysis	
L	Standard lapse rate $-1.98^{\circ}\text{C}/1000 \text{ ft}$	deg/ft
m	Slope of a line at a point	
m	Mass	slug
mac	Mean aerodynamic chord	
M	Mass, dimensional analysis	
M	Mach, flight or free stream	
$M_i$	Indicated Mach	
$M_{ic}$	Indicated Mach corrected for instrument error, $M_i + \Delta M_{ic}$	
$\Delta M_{ic}$	Machmeter instrument correction	
$\Delta M_p$	Machmeter position error corresponding to $P_p$	

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>ARABIC</u> <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$\Delta M_{pc}$	Machmeter position error correction	
M	Moment	ft lb
MAC	Mean aerodynamic chord	
n	Load factor	
n	Number of stages	
N	Nozzle	
N	Revolutions per minute	
NACA	National Advisory Committee for Aeronautics	
NASA	National Aeronautics and Space Administration	
$N_{pr}$	Prandtl number	
P	Power	hp, ft lb/sec
P	Pressure, general	lb/in <sup>2</sup>
P	The applied pressure at a point at a time, t	in Hg
$P_a$	Atmospheric pressure corresponding to $H_c$	in Hg

### SYMBOLS, TERMS, AND ABBREVIATIONS

ARABIC <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$P_{a_{sl}}$	Atmospheric pressure at standard sea level	2116.22 lb/ft <sup>2</sup> 29.92126 in Hg
$P_i$	The indicated pressure at a point at a time, t	in Hg
$\Delta P_p$	Static pressure error or position error	in Hg
$P_s$	Pressure corresponding to $H_{ic}$	in Hg
$P_s$	Specific Excess power	
$P_t$ or $P_T$	Free stream total pressure	in Hg, lb/in <sup>2</sup>
$P'_t$	Total pressure at total pressure source	in Hg
PE	Potential energy	ft lb
$q$	Dynamic pressure, $\rho V_T^2/2$	in Hg
$q_c$	Differential pressure, $P'_t - P_a$	in Hg
$q_{cic}$	Differential pressure corresponding to $V_{ic}$ , $P'_t - P_s$	in Hg
Q	Heat or heat energy	btu
$\Omega$	Torque	in lb

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>ARABIC</u> <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
r	Blade length	in, ft
R	Radius of turn	
R	Range	
R	Gas constant for dry air	ft <sup>2</sup> /sec <sup>2</sup> °R
°R	Degrees Rankine	deg
R <sub>e</sub>	Radius of the earth	ft
R <sub>e</sub>	Reynolds Number	
RF	Range factor	
ROC	Required operational capability	
ROC	Rate of climb	
RW	Relative wind	
s	Specific Entropy	btu/lb
s	Distance	ft
S	Total wing or planform area	ft <sup>2</sup>
S <sub>a</sub>	Air distance	ft

**SYMBOLS, TERMS, AND ABBREVIATIONS**

**ARABIC**

**Symbol or Term**

**Definition**

**Units**

$s_g$

Ground roll distance

ft

SM

Stall margin

SR

Specific range

nm

SFC

Specific fuel consumption

SPR

Stage pressure ratio

t

Thickness

in, ft

t

Time

sec

$t_a$

Atmospheric temperature

°C

$t_{as}$

Standard day atmospheric temperature corresponding to  $H_c$

°C

$t_{a_{sl}}$

Standard sea level atmospheric temperature

15°C

$t_{at}$

Test day atmospheric temperature

°C

$t_i$

Indicated temperature

°C

$t_{ic}$

Indicated temperature corrected for instrument error,  $t_i + \Delta t_{ic}$

°C

$\Delta t_{ic}$

Air temperature instrument correction

°C

SYMBOLS, TERMS, AND ABBREVIATIONS

ARABIC <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
T	Temperature	deg
T	Time, dimensional analysis	
T	Turbine	
T	Propeller thrust	lb
$T_a$	Atmospheric temperature	$^{\circ}$ K
$T_{as}$	Standard day atmospheric temperature corresponding to $H_c$	$^{\circ}$ K
$T_{a_{sl}}$	Standard sea level atmospheric temperature	$288.16^{\circ}$ K
$T_{at}$	Test day atmospheric temperature	$^{\circ}$ K
$T_i$	Indicated temperature	$^{\circ}$ K
$T_{ic}$	Indicated temperature corrected for instrument error, $T_i + \Delta T_{ic}$	$^{\circ}$ K
$\Delta T_{ic}$	Air temperature instrument correction	$^{\circ}$ K
$T_t$	Total temperature	$^{\circ}$ K
$T_T$	Total temperature (general)	deg

SYMBOLS, TERMS, AND ABBREVIATIONS

ARABIC <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
TE	Total energy	
THP	Thrust horsepower	
TIT	Turbine inlet temperature	deg
TPR	Total pressure ratio	
TSFC	Thrust specific fuel consumption	lb/hr
u	Linear velocity	ft/sec
v	Velocity or true airspeed	
$v_c$	Calibrated airspeed, $v_i + \Delta v_{ic} + \Delta v_{pc}$	kts
$v_e$	Equivalent airspeed, $v_c + \Delta v_c$ or $v \sqrt{\sigma}$	kts
$v_i$	Indicated airspeed	kts
$v_{ic}$	Indicated airspeed corrected for instrument error, $v_i + \Delta v_{ic}$	kts
$\Delta v_{ic}$	Airspeed indicator instrument correction	kts
$v_{ic_l}$	Indicated airspeed corrected for instrument and lag errors,	
	$v_i + \Delta v_{ic} + \Delta v_{ic}$	kts

SYMBOLS, TERMS, AND ABBREVIATIONS

<u>ARABIC</u> <u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$\Delta V_{ic_l}$	Airspeed indicator lag corrections	kts
$\Delta V_p$	Airspeed indicator position error corresponding to $\Delta P_p$	kts
$\Delta V_{pc}$	Airspeed indicator position error correction	kts
$\Delta V_c$	Compressibility correction	kts
$V_s$	Standard day true airspeed	kts
$V_t$	Test day true airspeed	kts
w	Relative velocity	ft/sec
w or W	Work	ft/lb
w	Downwash velocity	ft/sec
W	Aircraft gross weight	lb
$\dot{w}_a$	Airflow rate	lb/hr or lb/sec
$\dot{w}_f$	Fuel flow rate	lb/hr or lb/sec
W.L.	Water line	
x	Distance	ft
z	Energy reference height	ft
$\propto$	Proportional to	

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$\alpha$	Angle of attack	deg, rad
$\beta$	Angle of sideslip	deg
$\beta$	Bypass ratio	
$\gamma$	Ratio of specific heats	
$\gamma$	Flight path angle	deg
$\delta$	Pressure ratio , $P_a/P_{a_{sl}}$	
$\delta_{ic}$	$P_s/P_{a_{sl}}$	
$\delta_L$	Laminar boundary layer thickness	
$\delta_T$	Turbulent boundary layer thickness	
$\delta$	Wedge angle or turning angle	
$\Delta$	Change in any quantity	
$\epsilon$	Axial strain	
$\epsilon$	Downwash angle	deg, rad
$\eta$	Efficiency	
$\eta_o$	Overall efficiency	
$\eta_p$	Propulsive efficiency	

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$\eta_{th}$	Thermal efficiency	
$\theta$	Temperature ratio, $T_a/T_{a_{sl}}$	
$\theta_s$	$T_{as}/T_{a_{sl}}$	
$\theta_t$	$T_{at}/T_{a_{sl}}$	
$\theta$	Shock wave angle	
$\lambda$	Lag constant	sec
$\lambda_{H_{ic}}$	Lag constant corresponding to $H_{ic}$	sec
$\lambda_s$	Static pressure lag constant	sec
$\lambda_{s_{sl}}$	Lag constant at standard sea level	sec
$\lambda_{s_{sl}}$	Static pressure lag constant at standard sea level	sec
$\lambda_t$	Total pressure lag constant	sec
$\lambda_{t_{sl}}$	Total pressure lag constant at standard sea level	sec
$\lambda$	Taper ratio	
$\lambda$	Sweep angle	deg

SYMBOLS, TERMS, AND ABBREVIATIONS

<u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$\nu$	Coefficient of absolute viscosity	$\text{lb sec/ft}^2$
$\mu$	Viscosity at temperature $T_a$	$\text{lb sec/ft}^2$
$\mu_{H_{ic}}$	Viscosity corresponding to $H_{ic}$	$\text{lb sec/ft}^2$
$\mu_{sl}$	Viscosity at standard sea level	$3.7452 \times 10^{-7}$ $\text{lb sec/ft}^2$
$\mu$	Mach angle	deg
$\mu$	Coefficient of friction	
$\nu$	Kinematic viscosity	ft sec
$\nu$	Turning angle	deg
$\pi$	3.14159 ...	
$\pi$	Buckingham $\pi$	
$\rho$	Air density	$\text{slug/ft}^3$
$\rho_a$	Standard day air density corresponding to $H_c$	$\text{slug/ft}^3$
$\rho_{sl}$	Air density at standard sea level	.0023769 $\text{slug/ft}^3$
$\rho_t$	Test day air density	$\text{slug/ft}^3$
$\sigma$	Density ratio, $\rho_a/\rho_{sl}$	
$\sigma_s$	$\rho_s/\rho_{sl}$	

**SYMBOLS, TERMS, AND ABBREVIATIONS**

<u>Symbol or Term</u>	<u>Definition</u>	<u>Units</u>
$\sigma_t$	$\rho_t/\rho_{sl}$	
$\sigma$	Axial stress	lb/in <sup>2</sup>
$\sigma$	Solidity ratio	
$\tau$	Acoustic lag	sec
$\tau$	Shear stress	lb/in <sup>2</sup>
$\phi$	Bank angle	deg
$\omega$	Rate of turn	deg/sec or rad/sec

SUBSCRIPTS

<u>Symbol or Term</u>	<u>Definition</u>
a	Ambient
a	Available
cr	Critical
e	Equivalent
ex	Excess
f	Final
i	Induced
i	Initial
iw	Corrected to a standard weight
L	Laminar
M	Wave
N	Normal (perpendicular)
o	Stagnation or total
p	Parasite

SUBSCRIPTS

<u>Symbol or Term</u>	<u>Definition</u>
p	Pressure
r	Required
r	Root
s	Static
s	Standard day
sl	Sea level
t	Tangential
t	Test day
T	Total
TD	Touchdown
TO	Takeoff
X	Conditions upstream of shock wave
Y	Conditions downstream of shock wave
OL	Zero lift
1,2,3, etc.	Specific condition or station

SUBSCRIPTS

Symbol or Term

Definition

Free stream condition

SUPERSCRIPT

Symbol or Term

Definition

Choked condition